



#### Outline



- F-22 Structural Materials Mix
- Design Allowables Development
- Factors Affecting Material Properties
- Maintaining Allowables in Production
- Summary
- Questions





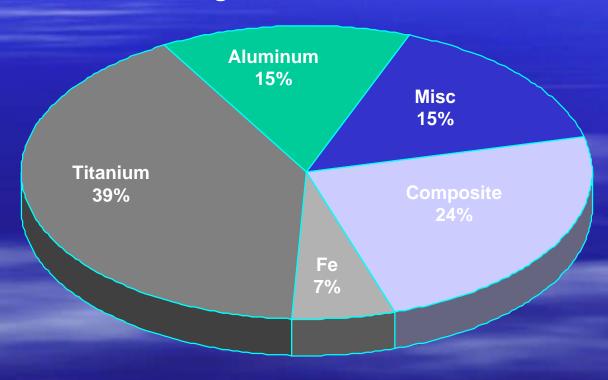
### F-22 Structural Materials Mix



### F-22 Materials Mix



#### **Airframe Structural Weight**







# Design Allowables Development



# Design Allowables Development

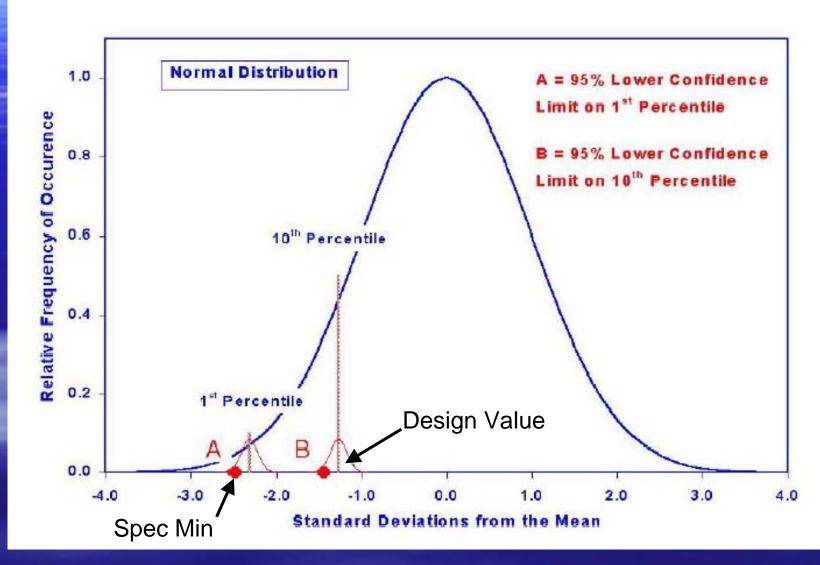


- Structural Development Test (SDT) Pgm
  - Material Spec Development
    - Spec Min Values Estimated
  - Mechanical Property Development
    - ~3,600 Titanium Coupon Tests
      - Static
      - Durability & Damage Tolerance (DADT)
    - Statistically Reduced Allowables Developed
      - "B" Basis
      - Spec Values Verified
    - DADT Design Properties Established



### Design vs Spec Values

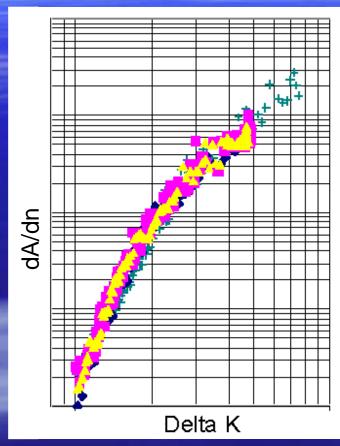




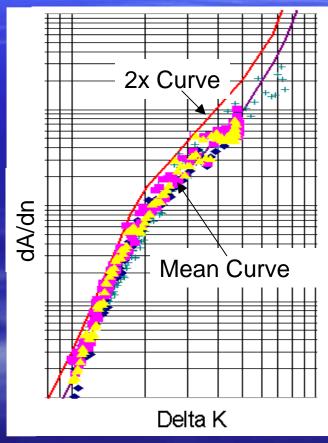


### Crack Growth Rate





Collect Crack Growth Data



Develop Mean & 2 x Rate Curves





# Factors Affecting Material Properties



# Factors Affecting Mechanical Properties



**Raw Materials** 

Source Mfg Process Quality Morphology Consolidation

Electrode Fab Melt Practice

Conversion

Ingot to Product TMP Route

**Heat Treatment** 

Solution Heat Treat Temp
Dwell Time
Cooling Rate
Age or SR Cycle





### Maintaining Allowables in Production



### Maintaining Allowables

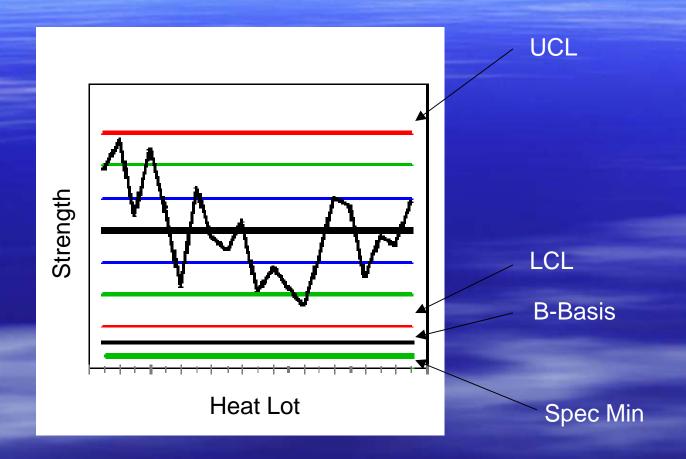


- Existing Processes
  - Establish Fixed Practice Ageements (FPAs)
  - -Monitor
    - SPC
- Qualify New Processes
  - Insure Design Allowables Supported
    - Static
    - DADT



### SPC







### New or Changed Process



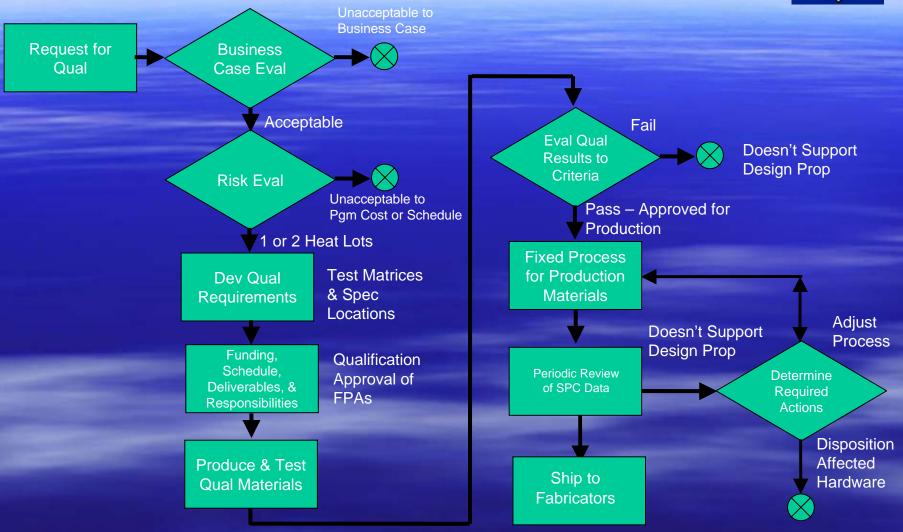
- Insure that new vendors &/ or processes:
  - Warrant expenditure of program & vendor resources
  - Do not increase the program risks
  - Support existing design properties
  - Maintain consistent properties when transferred to production



### **Qualification Flow**



MRB Disposition





### Acceptance Criteria



Test Type	Acceptance Criteria	Comment
Chemistry	Material Spec	
Micro & Macro Structure	Material Spec	
Tensile RT RT Exposed 700 F	Support "B" Support "B" Consistent w SDT	No Loss in properties Loss consistent with F/A-22 Database
Strain Life	No Results < 25% D.L. Geom Avg > 80% D.L.	Replicates at three strain levels
FCGR	Smoothed Data does not exceed 2X Curve	
Fracture Toughness	Material Spec	
Stress Corrosion	Material Spec	



### Summary



- F-22 structural integrity is based on production materials supporting the design properties
- Multiple factors can affect material properties
- Need to monitor existing processes
  - Detect detrimental trends & take corrective actions
- New process / vendor qualification
  - Support design values
    - Static
    - DADT





## Questions