

# Titanium Trends and Usage in Commercial Gas Turbine Engines

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# What is the Future of Aerospace Titanium?



- Fuel represents ~50% Of airline costs
  - Engines provide significant opportunity for fuel burn reduction
  - Composite airframes also contribute to performance improvements
- Improved efficiency objectives are driving cycle and architecture requirements challenging Ti engine usage
- Airframe applications driven by composite compatibility promoting Ti usage

Ti research focus – affordability then performance ...is it the right balance?

#### **Outline**

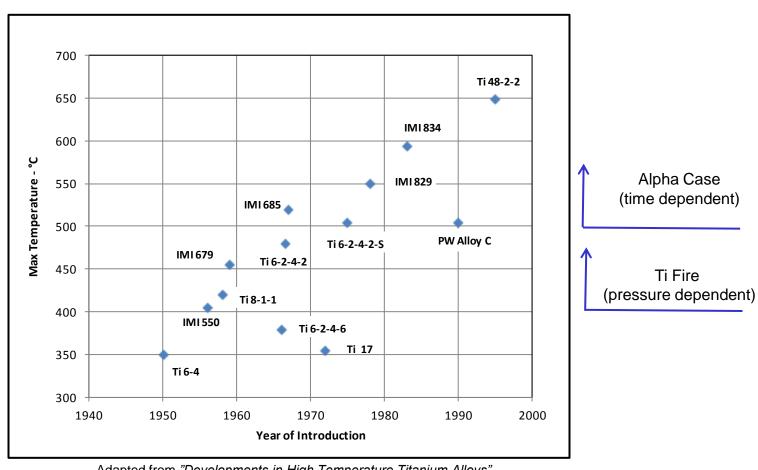


- ➤ Historic Titanium Usage
- ➤ Trends in Engine Architecture & Cycle
- ➤ Impact to Material Usage
- ➤ The Challenge Material Requirements for Next Generation Engines

### **Historic Titanium Usage**



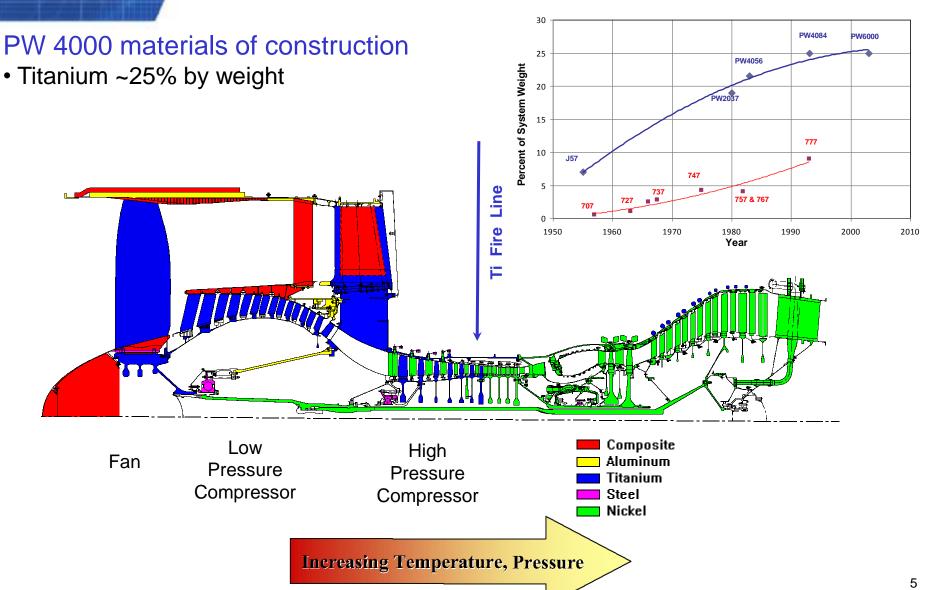
Titanium historically limited by Alpha Case and compressor fire potential (& cost)



Adapted from "Developments in High Temperature Titanium Alloys" Blenkinsop, P.A. Titanium Science & Technology 1984

## Titanium Usage – Last Century





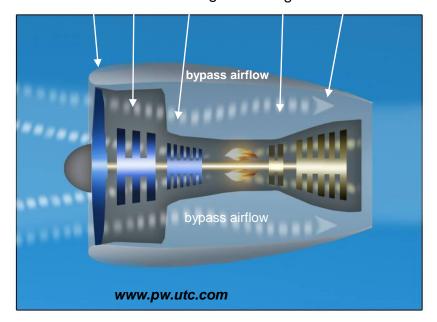
### **Engine Efficiency Drivers**



#### Overall efficiency = Propulsive efficiency x Cycle efficiency

#### Conventional Turbofan

Fan Compressor Turbine Low High High Low



Bypass Ratio = Bypass Airflow / Core Airflow

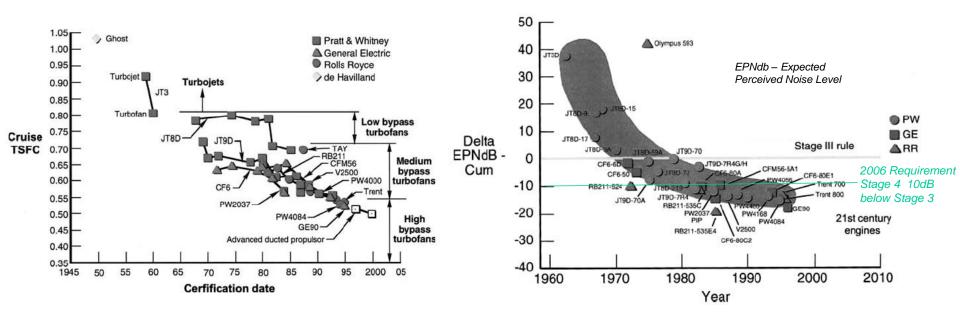
- Propulsive efficiency: Driven by increased bypass ratio
  - Higher bypass ratio → larger diameter fan
  - To maximize benefits of a large fan:
    - Need hollow titanium or composite fan blades
    - Need light weight composite static structure
- <u>Cycle efficiency</u>: Driven by higher pressure ratio and higher turbine inlet temperature
  - Higher pressure ratios require higher temperature capability in the high pressure compressor
  - Higher turbine inlet temperatures require higher temperature turbine materials and coatings.

# Bypass Ratio Drives Efficiency & Noise



#### As bypass ratios increases

- Thrust Specific Fuel Consumption (TSFC) and noise decrease



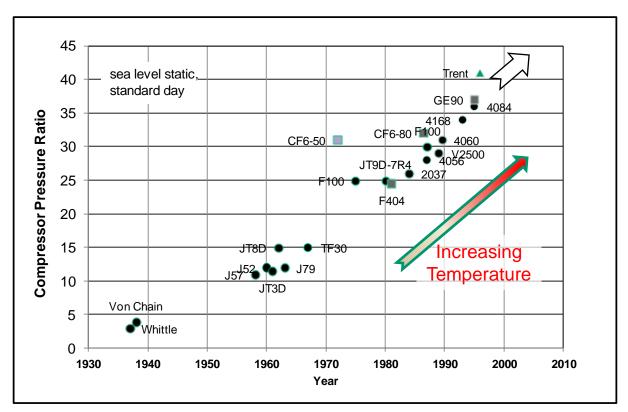
Gas Turbine Technology Evolution: A Designer's Perspective

Bernard L. Koff *TurboVision, Inc., Palm Beach Gardens, Florida 33418* JOURNAL OF PROPULSION AND POWER Vol. 20, No. 4, July–August 2004

### **Compressor Temperature Trend**



Engine efficiency increases as compressor pressure ratios increase. Compressor temperature increases with pressure.



"As the operating temperature of turbine engines increases, titanium will struggle to maintain its foothold in aircraft high pressure compressor disks."

- G. Vroman, Sr. VP, ATI-Ladish, from American Metal Market, 1998

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### Engine Changes – Material Impacts 💝 Pratt



#### Engine architecture and cycles driving material changes away from titanium:

- Bypass Ratios Increasing = Larger Diameter Fans
  - Organic matrix composites (OMCs) replacing Ti in fan blades and containment cases
- Compressor pressure ratios increasing = Hotter compressors and turbines
  - Transition to integrally bladed rotors results in material selected by gas path requirements
  - Nickel replacing titanium in earlier stages of the high pressure compressor (HPC)
  - High temperature turbines reduce stages where gamma alloys work
- · Core diameters are decreasing
  - The volume (size) of Ti & Ni components is decreasing

#### **Bypass Ratio & Fan Blade Materials**



2010s

Materials play a key role in enabling higher bypass ratios by enabling larger diameter light weight fans.

1990s Hollow Ti

PW 4084 BPR = 6.4

2000s Composite / Hybrid Metallic Composite



GE 90 BPR = 9



PW 1524G BPR = 12

JT3D BPR = 1.3

1960 - 1980s Solid Ti

#### **Bypass Ratio & Fan Case Materials**



Solid steel and titanium fan containment cases being replaced by Kevlar composite and all composite designs to reduce weight.

2010s

#### 1980s



PW 4056 BPR = 4.8 Solid Steel

#### 1990s



V2500 BPR = 4.5 - 5.4 Solid Titanium

#### **2000s**



GP7200 BPR = 8.8 Kevlar/Aluminum

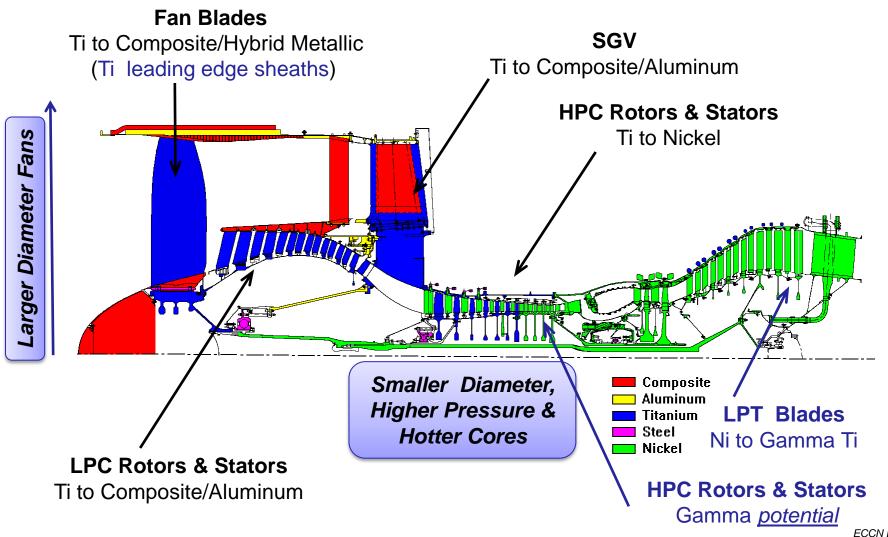


PW1524G BPR = 12 Composite

### **Engine Changes**



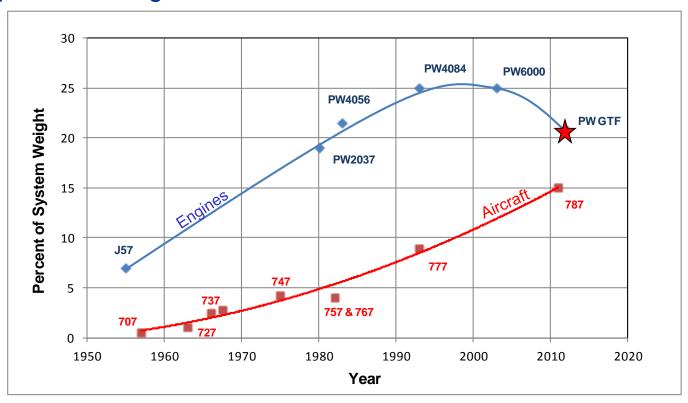
#### Engine architecture/cycle drives material changes



## **Titanium Usage – This Century**



- Temperature limitations Ceiling for utilization in engines
- New engines Higher bypass ratios, smaller cores, increased temperature / larger Fans → reduced Ti content

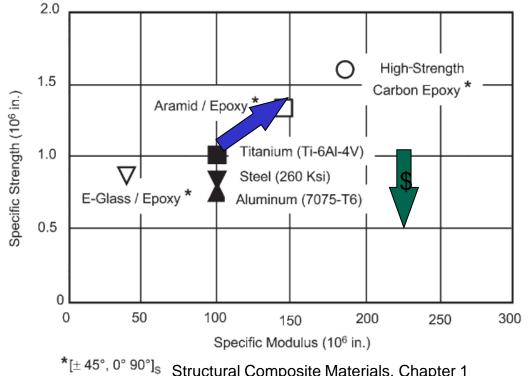


Commercial aircraft and engines

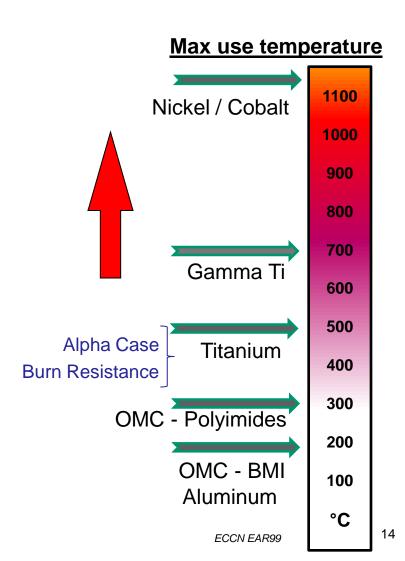
# Ti Development Challenges...From an Engine Perspective



- Cost (always of great interest)
  - Compete with Al & steel
- High specific capability (propulsive efficiency)
  - Compete with composites
- Environmental resistance (thermal efficiency)
  - Compete with superalloys



ls Structural Composite Materials, Chapter 1 F.C. Campbell, 2010, ASM International



## **Airframe Technology Examples**



#### Aluminum industry's response to composites: increase performance

- High strength corrosion resistant alloys
  - Increased strength & design compatible
- 3<sup>rd</sup> Generation Al-Li alloys
  - Increased specific strength & stiffness
- Novel manufacturing methods (FSW)
- Hybrid materials (GLARE)
  - GLARE® (GLAss fiber REinforced aluminum)



Similar advances/approach required for Titanium

# Next Generation Engine Opportunities



- Fan & LPC Specific strength vs. OMCs
  - Cost Effective MMCs, Ti + B (P/M), Hybrid Ti / OMCs
- HPC Temperature capability and burn resistance
  - Higher Temperature Capable & Burn Resistant Alloys and Coatings, Expanded Use of Gamma Ti
- LPT Creep and oxidation (Gamma Ti)
  - Higher Temperature Capable Alloys / Coatings
- Low cost material and manufacturing processes
  - Low Cost Raw Materials Ti Reduction, Melting, Conversion
  - Additive Manufacturing to Enable Hybrid & Low Cost Solutions

## Summary



- Titanium usage decreasing in engine applications
  - Driven by cycle and architecture changes
  - Displaced by organic matrix composites and super alloys
- Aluminum industry response to composite threat is increased performance / advanced fabrication technologies
- Technology opportunities for Titanium
  - Increased specific capability
  - Advanced manufacturing for high performance structures
  - Improved temperature capability systems (coatings)
  - Continued work on cost reduction

# **Thank You**



